



LES of a simplified HVAC system used for aero-acoustic predictions

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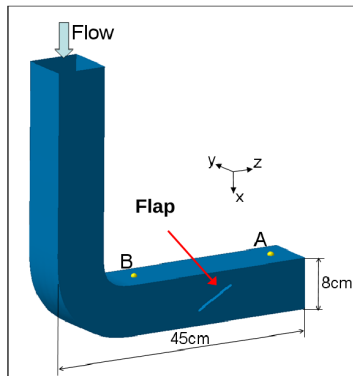


- 1 Heat and Ventilation Air Conditioning: HVAC
 - Introduction
 - Flow Features
 - Acoustic
- 2 Paraview in a Visualization Cluster (Placement of E. Harrison)
- 3 Conclusions and Future Work



HVAC

- HVAC stands for Heat and Ventilation Air Conditioning and it is a common system in several engineering application
- This simplified configuration has been proposed by German car industry for validation of numerical methods
- The system is composed by a duct bend with a flap
- The duct creates a jet in an open space
- Both experimental data for fluid flow (PIV) and aeroacoustic are available





Test case definition

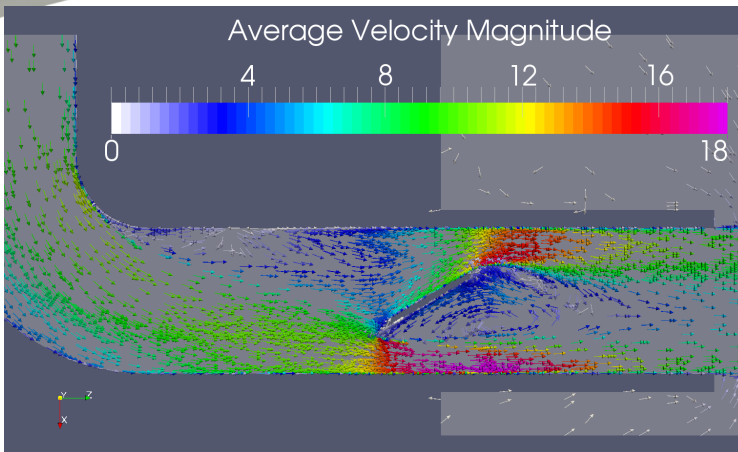
- LES with Smagorinsky model
- Standard air at 15° C
- Inlet based on a fully developed duct flow computed with RANS model (4 eqs v^2f) with DF-SEM for turbulent fluctuations
 - Bulk velocity $U_B = 7.5 \text{ m/s} \Rightarrow Re \approx 40000$
 - Low-Re Reynolds stress model also tested but flow is re-laminarize
- Free inlet/outlet BC at the side and at the exit of the plenum
 - BC based on the Bernoulli relation between the face and a point on the same stream-line place at ∞ in case of incoming flow
 - Homogeneous Neumann is applied on the velocity
 - Assuming $\underline{u}_\infty = 0$ the dynamic pressure at the face is:

$$p_f = -\frac{1 + K}{2} \rho_f \underline{u}_f \cdot \underline{u}_f$$

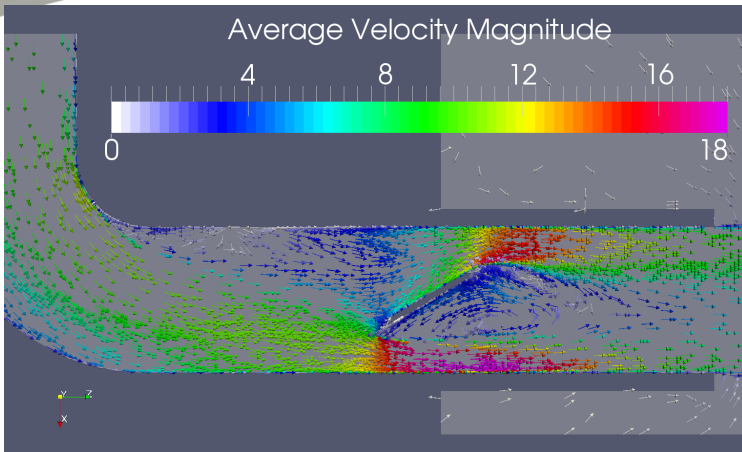
with K being a head loss.

(Loading movie)

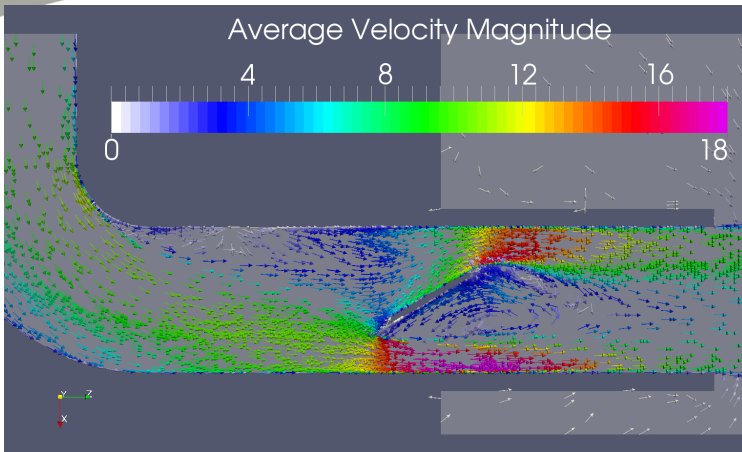
Figure 1: Iso-surfaces of Vorticity.



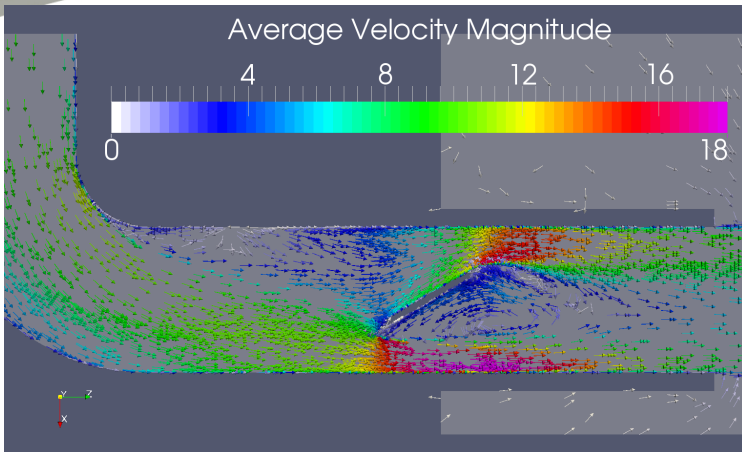
- Flow separation after the bend
- Impingement of the separated flow on the obstacle
- Recirculation behind the obstacle
 - Two counter-rotating vortices



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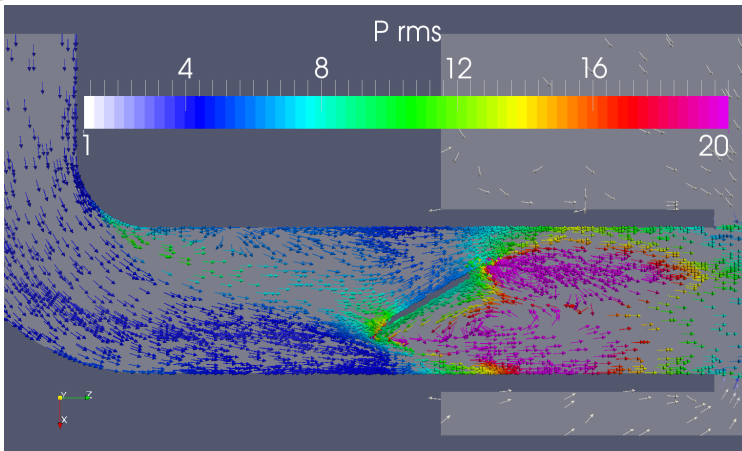
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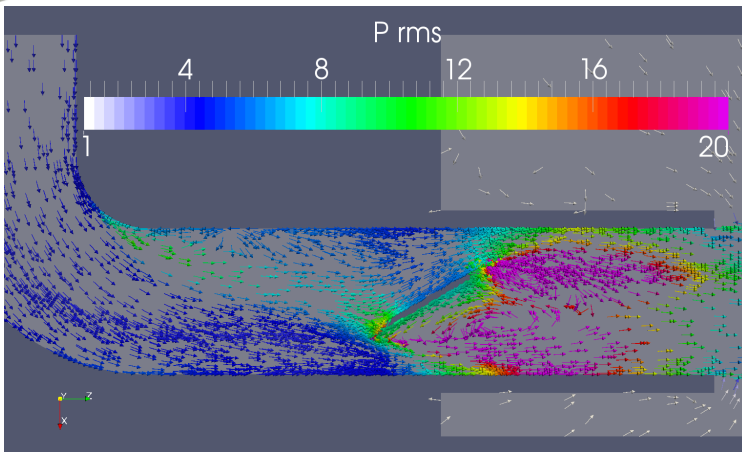
Pressure RMS (Mid Plane)



- Pressure fluctuations are peaking in recirculation areas
- These areas are candidate to be the location of the noise sources



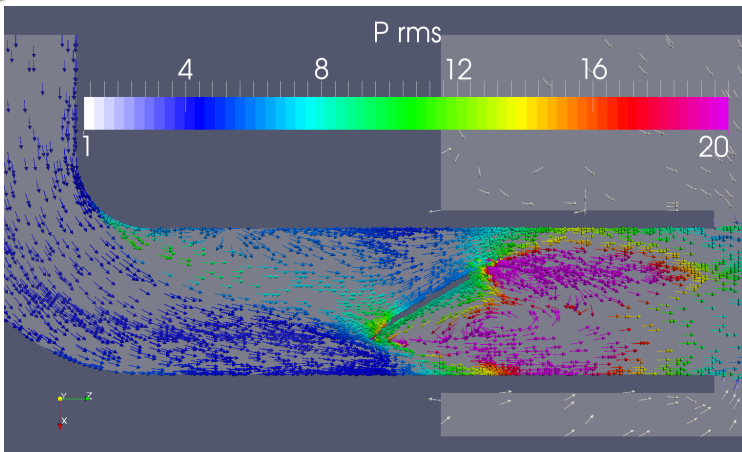
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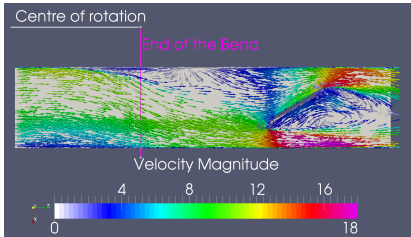
Pressure RMS (Mid Plane)



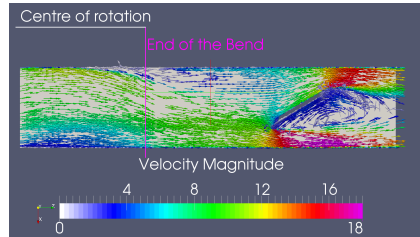
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XZ plane (obstacle plane)

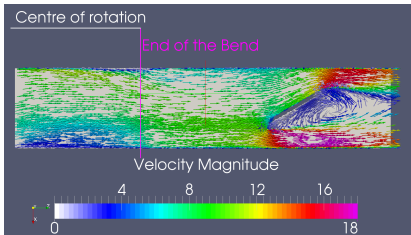
Mid Plane



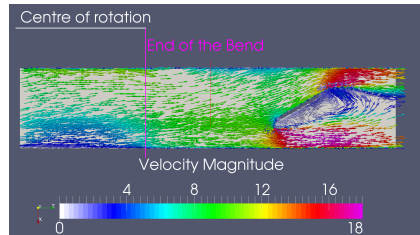
$2y/D_H = 0.5$



$2y/D_H = 0.75$

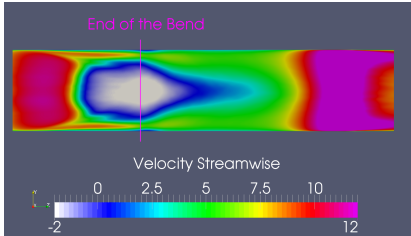


$2y/D_H = 0.85$

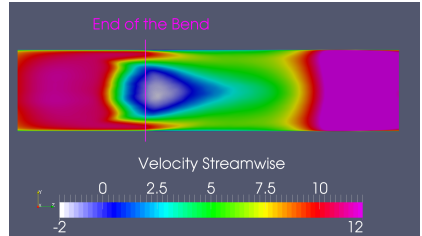


YZ plane (View from bending centre)

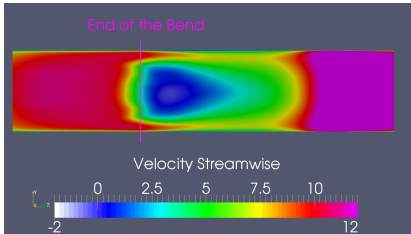
Top wall $x/D_H = 0.01$



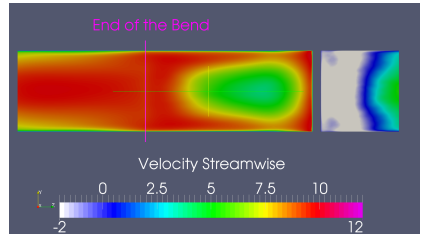
$x/D_H = 0.06$



$x/D_H = 0.125$

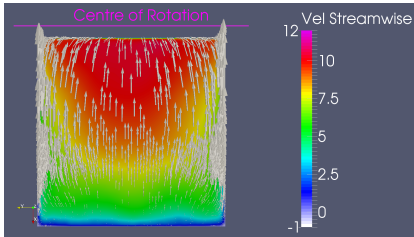


$x/D_H = 0.375$

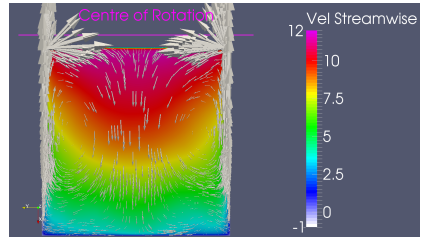


XY plane (Streamwise)

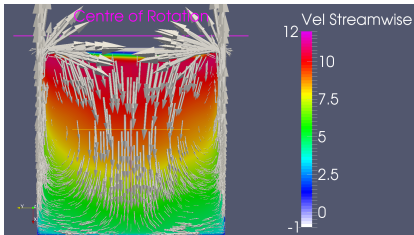
Inlet bend $\theta = -90^\circ$



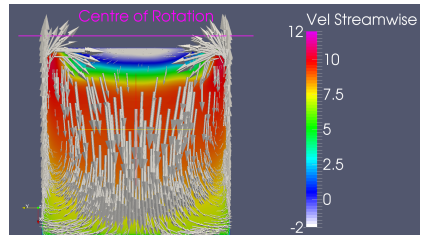
$\theta = -60^\circ$



$\theta = -30^\circ$

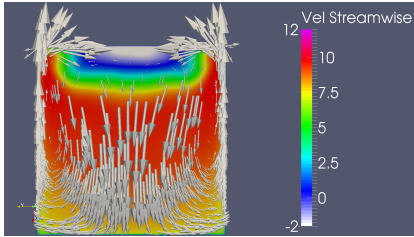


Outlet bend $\theta = 0^\circ$

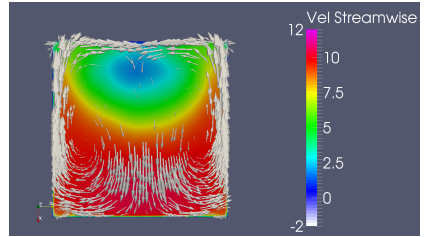


XY plane (Streamwise)

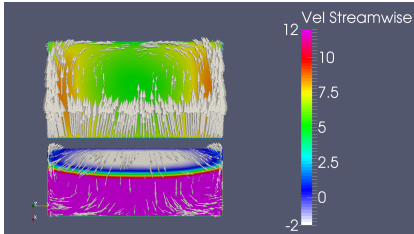
Near outlet bend $z/D_H = 0.125$



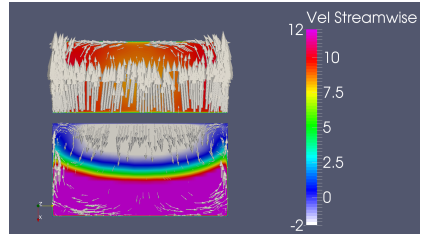
$z/D_H = 1.0$

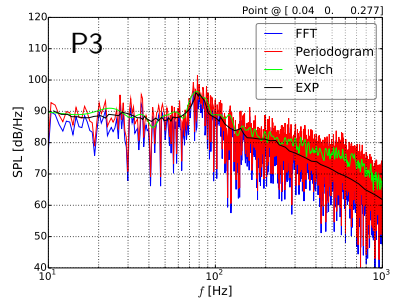
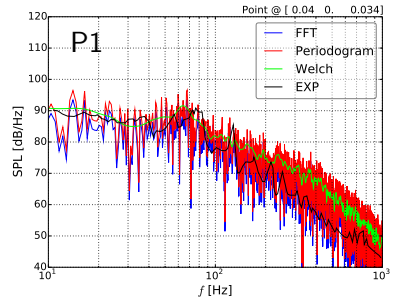
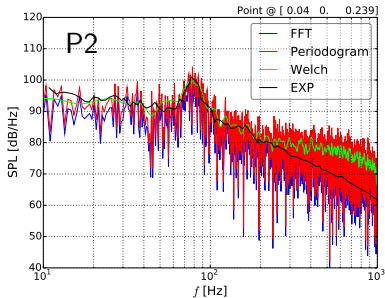
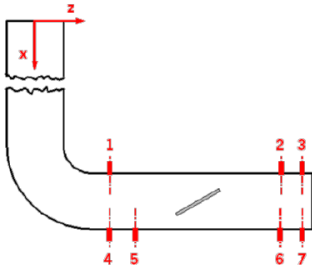


$z/D_H = 1.75$



$z/D_H = 2.0$





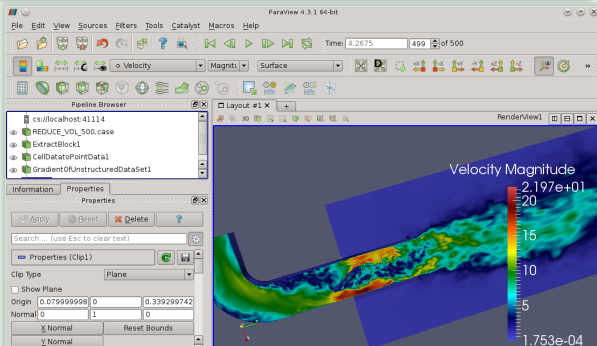


Why Paraview?

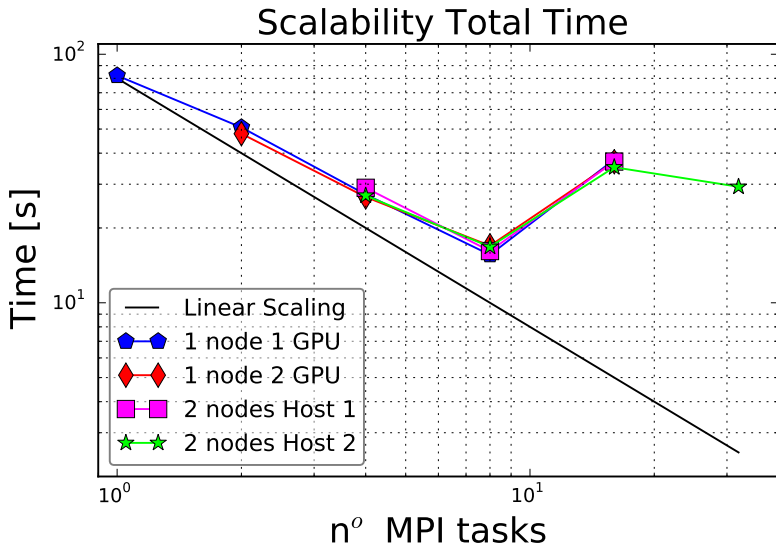
- It is not always possible to use co-processing (i.e. Catalyst)
- Possible to use remote machines with Client/Server mode
- We have a visualization cluster: maybe we can use it:
 - 2 fat nodes with 32 computing nodes and 64GB RAM
 - 2 GPUs attached to each node

The Pipeline

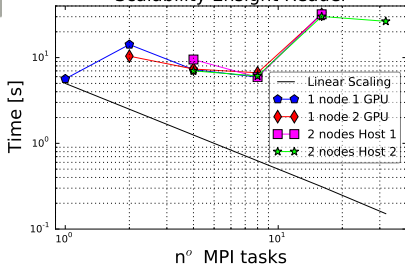
- Load of the data \Rightarrow **EnightReader**
- Convert data from cells to nodes \Rightarrow **CellToPoint**
- Gradient \Rightarrow **GradUnstructuredMesh**
- Update the layout \Rightarrow **GeoRepresentation**



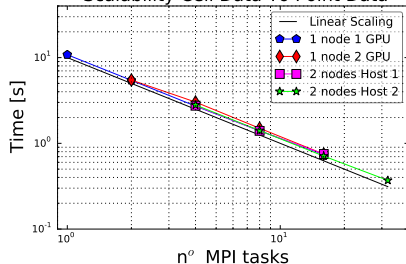
Scalability Pipeline in loading a time step



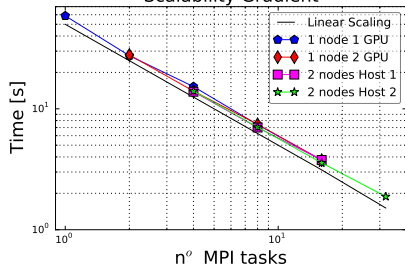
Scalability Enight Reader



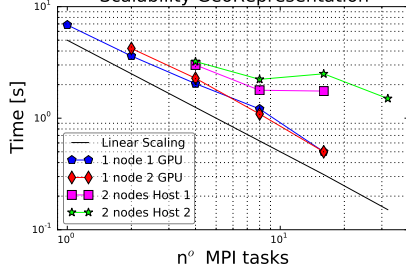
Scalability Cell Data To Point Data



Scalability Gradient



Scalability GeoRepresentation





Conclusions and Future Work

HVAC

- LES of a simplified HVAC has been presented and the available results are in agreement with experimental data available
- Application of Client/Server Paraview mode has been shown and poor scalability of the Ensign reader has been identified

Future work

- Further comparison with experimental value (hydrodynamic)
- Further investigations into aeroacoustic using the Curle's analogy
- More calculations with different SGS models are on going
- Awarding of a 1 year project to implement a FWH aeroacoustic module into *Code_Saturne*